Reliable Products and Structures

John Orr Lecture

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Outline

- What constitutes reliability?
- Systems analysis and characterisation
- Causes of failure
- Achieving reliability
- Conclusions
- Case studies in failure analysis



Reliability Requires Communication

















Reliability rests on a long history of scientific and engineering achievement





The quality and reliability of its phones in rural areas led Stromberg Carlson to became known as "the farmer's friend"







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Point Pleasant bridge over Ohio River, West Virginia, December 15 1967







Point Pleasant bridge over Ohio River, West Virginia, December 16 1967

- Bridge built 1920's from new high strength steel (UTS = 840 MPa)
- Design stress = 350 MPa, Eye-bar $k_t = 2.7$
- Fracture from 0.12" (3 mm) crack in tie-bar 55' long, 2" thick, 27.5" across eye
- Paint on fracture surface



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- Crashed shortly after take-off at Guam
- Four flight control system computers translate cockpit input into control surface movement
- Moisture distorted air-pressure readings
- Re-calibrated by maintenance crew – then evaporated
- B2_Bomber_Crash_Guam.flv (1:50)
- Pressure differences were miniscule, but they were enough to confuse the FCS
- FCS took control; triggering a premature takeoff, automatically driving the aircraft into a 30-degree, noseup pitch and overruling the pilot's efforts to regain control





Ariane 5, 40 seconds later 1996

- Ariane 5 launch vehicle failed on its maiden flight in June 1996
- About 40 seconds after lift-off the • launcher veered off its flight path, broke up and exploded
- High aerodynamic loads due to an ۲ angle of attack $> 20^{\circ}$ led to separation of the boosters from the main stage, in turn triggering the self-destruct system of the launcher

Software problem arose from a numeric overflow in the Inertial Reference System program during conversion of a 64-bit to a 16-bit number

The piece of code had been originally written for the Ariane 4 and was reused in the Ariane 5

The bug caused both Inertial Reference Systems to crash

Ironically, the code was superfluous once lift-off occurred







- 'Cabriolet' conversion shortly after take-off at 350 mph and 24,000 feet
- Everyone belted in their seat survived
- Illustration of damage-tolerance by fuselage of a 'two-bay' crack
- Multi-site corrosion-fatigue cracks at rivet holes on lap joint
- Scrim cloth and adhesives absorbed water and became brittle
- Poor maintenance inspection record
- Passenger noticed a 120 mm crack on embarkation

Aloha737.mov







- Strong crosswind as storm cell passes across Germany
- Gusts up to 50 knots (93 km/h)
- Illustration of damage-tolerance of wing under impact conditions
- Just before the final touchdown, there was a gusting crosswind from the side
- Aircraft's left wing-tip struck the runway
 - Bends wing-tip fence
 - Causes minor damage to the wing surface
- Crew aborted landing
- Lufthansa A320 Movie



Depends on viewpoint – user or manufacturer

- User would like zero failures hence warranty
- Manufacturer accepts a level of failure
 - Design for a statistically reliable failure rate
 - Put in place quality assurance systems
 - Learn from history
- Some SME's do not explicitly consider likelihood and consequences of failure – e.g. bidet



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Define a reliable product or structure as:

One that operates in the expected way, for the anticipated life and with the estimated costs



A product or structure that is characterised by:

- Maximised return on investment
 - Cost–optimised manufacture/fabrication, operation and retirement develop for reliability





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A product or structure that is characterised by:

- Maximised return on investment
 - Cost–optimised manufacture/fabrication, operation and retirement develop for reliability
 - Requires organisations to become knowledgesharing enterprises
 - Capacity for making explicit the implicit knowledge of employees
 - Best practices
 - Lessons learned
 - Boundary conditions and short cuts
 - Ability to continually develop the knowledge base through fast learning and innovation
 - Proactive real-time knowledge management



A product or structure that is characterised by:

- Achievement of specified service reliability
 - Probabilistic risk assessment
 - Integrate diverse aspects of design and operation in order to assess the risk
 - Appropriate modelling and testing
 - Structural health monitoring through built-in test equipment (BITE)
- Defect-tolerance and fail-safe operation
 - Appropriate inspection intervals and procedures
 - Alternate load paths and critical systems

Complication: Many products and structures are actually complex systems \longrightarrow difficult analysis

Need tools to characterise systems

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 Complex systems require partitioning and concurrent consideration





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- Have opportunity for innovation at each stage of the analysis
 - Explicitly consider material/design interactions
 - Seek opportunities from innovation in materials or fabrication/manufacturing





- Probabilistic reliability/risk assessment of systems and sub-systems involves causeresult analysis
 - Integrate diverse aspects of design and operation in order to assess the risk
 - Develop a set of possible casualty sequences and determine their outcomes
 - System models consist of event trees and fault trees

Diagram from Japan Science and Technology Agency website



Systems models to achieve service reliability

- Event trees
 - Depict initiating events and combinations of system successes and failures
 - Event sequence analysis maps how the outcome of a desired objective (e.g. a flight) depends on the performance of critical systems (A-E)



Sum of event sequence failure probability gives aircraft accident probability

 \equiv <1 death in 10⁸ passenger \sqrt{V}^{E} miles

Sample event tree –

event sequence analysis for cable bus



Systems models to achieve service reliability

- Fault trees
 - Depict ways in which the system failures represented in the event sequences can occur



- Hierarchical logic model is developed
 - Defines all of the combinations of system, subsystem, component, and subcomponent faults that will result in the top event failure

Sample fault tree – system failures leading to loss of rudder control



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Systems models to achieve service reliability

- Failure mandalas* defined by the JST Agency
 - Depict hierarchical relationship between the elements that make up the components of failure

 $CAUSE \longrightarrow ACTION \longrightarrow RESULT$

- Core concept is at heart of the mandala
- First and second ring elements are applicable to any field
- Third level elements are intended to be field specific





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The "flow" down from Cause through Action to Result for any particular scenario can be easily understood and communicated

Include:

- Inappropriate design, fabrication or manufacture
- Inadequate management of the design process
- Human-system interactions which are not fail-safe
 - Flawed operational decision support systems
 - Dangerous sequences of actions
 - Inappropriate inspection or maintenance procedures
- Software and firmware problems
- System and structural complexity
 - New technology
- Human error

Inappropriate design, fabrication or manufacture
Liberty ships - 1942

New fabrication technology (welding) for traditional design (rivets)

Inadequate management of the design process

Alexander Kielland oil accommodation platform – delivered July 1976, failed March 1980

 Uncontrolled design changes late in partitioning or integration stages

Human-system interactions which are not fail-safe

- Flawed operational decision support systems
 - Challenger space shuttle

Human-system interactions which are not fail-safe

- Dangerous sequences of actions
 - Chernobyl Nuclear Reactor

Non-routine test operation:

- Isolation of the emergency core cooling system – had this been operational it might have reduced the impact of the accident
- Operation at low power levels (<< 700 MW) led to known reactor control rod instability
- Operation of additional core cooling pumps, which reduced the level of water in the steam separator

Test performed by night shift

Poor knowledge of characteristics

System and structural complexity

- New technology
 - Comet airliner entered service in 1952 flying from London to Johannesburg
 - Between May 1953 and April 1954 3 aircraft broke up in the air
 - Withdrawn from service and extensive investigation undertaken
 - Full-scale structural testing (for first time)
 - Retrieval of pieces from bottom of Mediterranean
 - New technology introducing new load cases high altitude flight for turbojet engines requiring cabin pressurisation giving out-of-plane bending
 - Mismatch between service loads and fatigue test procedure
 - Comet Video.VOB

How do we achieve success?

- Integrated and concurrent systems design
- Advanced modelling and expert systems
- In-service monitoring of loads, strains and damage
- Sophisticated testing and materials characterisation
- Detailed understanding of behaviour of cracked bodies
 - Fracture mechanics
 - Crack growth mechanisms
 - Fatigue
 - Environmentally assisted cracking; SCC, LMAC etc
 - Creep
- Fracture control plans

Integrated and concurrent systems design

Evolution of military aircraft engine development process

F100-100 (CIRCA 1970)	F100-220 (CIRCA 1980)	F119-100 (CIRCA 1990)
PERFORMANCE FOCUS	BALANCED FOCUS BETWEEN PERF. & DURABILITY	APPLICATION TO ALL SUBSYSTEMS/COMPONENTS
MINIMAL ANALYTICAL UNDERPINNING	MATERIAL CHARACTERIZATION	EXPANSION OF PROCESS TO
ITERATIVE TAAF APPROACH (Tool, Applying and Eix)	INCREASED EMPHASIS ON	(PERF, OPER, ETC)
LIMITED INSTRUMENTED TESTING	ATTENTION TO ACTUAL ENVIRONMENT/USAGE	EXTENSIVE ANALYTICAL RIGOR IN DESIGN
NON-REPRESENTATIVE ENDURANCE TESTING (QT)	REPRESENTATIVE ENDURANCE TESTING (AMT)	COMPREHENSIVE ENVIRONMENT/RESPONSE CHARACTERIZATION
INSUFFICIENT ANALYTICAL & EMPIRICAL TOOL SET	DAMAGE TOLERANCE DESIGN APPROACH FOR SAFETY	PROOF/MARGIN TESTING
LITTLE ATTENTION TO LIFE MANAGEMENT	 CRITICAL COMPONENTS FULL LIFE TESTING OF MAJOR STRUCTURAL COMPONENTS 	DAMAGE TOLERANCE EXTENDED TO MISSION CRITICAL COMPONENTS
		EXTENSIVE COMPONENT, SUBSYSTEM, & SYSTEM LEVEL "SMART" TESTING
		PROCESS DEVELOPMENT & MATURATION IN EMD

Advanced modelling and expert systems

Finite element modelling, e.g. of heat flow during welding – can calculate residual stresses and predict performance (Corus)

Advanced modelling and expert systems

Evolution of CAD/CAM in military aricraft engine development

Sophisticated testing and materials characterisation

- Comet was first example of full scale testing of aircraft
- Early testing was based on 'limit-load' with 'factor of safety'

Wing testing on a BE2 biplane in 1912 using bags of lead shot

Fokker states "wood is a fatigue conditioned material"

Sophisticated testing and materials characterisation

Structural testing of Boeing 777

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static loads to the aircraft

Sophisticated testing and materials characterisation

Synchrotron diffraction measurement of residual strains

Measuring bi-axial residual strains in the fir tree root of an ESKOM steam turbine blade at the European Synchrotron Radiation Facility in Grenoble, France

Sophisticated testing and materials characterisation

- Aero-engine turbine blades
 - Impact damage bird strike

Rolls Royce Trent engine – titanium fan blades

Sophisticated testing and materials characterisation

- Aero-engine turbine blades
 - Fretting fatigue of blade root

Rolls Royce Trent engine – titanium fan blades

Sophisticated testing and materials characterisation

- Propane rail wagons subject to heating from fires
 - If a tank holding a pressure liquefied gas (such as propane) fails suddenly, then part of the contained liquid can boil violently producing an explosive effect
 - Interested in factors affecting length of time tank will resist fire FE modelling confirmed with full-scale testing

Work by AM Birk at Queen's University, Ontario

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Want fracture-safe and fatigue-reliable structures

- Fatigue: Process of crack initiation and growth by microplasticity in metal - (generally, a nonlinear constitutive response in a material leading to hysteretic energy loss)
- Fracture: Sudden, catastrophic collapse under a static load (certain cases), a steadily rising load, or as the final stage in fatigue

Ongoing problem areas:

- Single parameter K-characterisation of crack tip stresses
- Material/environment-induced crack tip shielding
- Incorporation of residual stresses into life prediction
- Global manufacturing and supply chain

Detailed understanding of behaviour of cracked bodies

- Fracture mechanics
- Crack growth mechanisms
 - Fatigue
 - Environmentally assisted cracking; SCC, LMAC etc
 - Creep
- Joint problems (welds)
- FM-based fatigue design
 - BS 7608 (Eurocode 3)
 - ✤ BS 8118 (Eurocode 9)
 - ✤ BS 7910 (PD 6493)

Assessing fracture control plan viability

- Probability of failure decreases as FCI increases (a)
- Fracture control cost increases with FCI both for design and in operation (b)
- Can plot a total cost curve (c)
- Higher cost structures justify a higher FCI

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Conclusions

- Our ability to design and build very complex and statistically reliable structures and products has increased dramatically over the last 30 years
- Driven by:
 - Advances in computing power and sophistication of electronics
 - Development of fracture mechanics and its application to fatigue crack growth
 - Systems engineering approaches and life cycle analysis
 - Integrated, concurrent design
 - Progress in materials science and engineering
- Mechanical engineers have been at the heart of many of these developments

Conclusions

Case Studies in Failure Analysis

